

# TRANSPA: A Code for Transient Thermal Analysis of a Single Fuel Pin

F. Coyne Prenger

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# TRANSPA: A CODE FOR TRANSIENT THERMAL ANALYSIS OF A SINGLE FUEL PIN

by

F. Coyne Prenger

## ABSTRACT

An analytical model (TRANSPA) for the transient thermal analysis of a single uranium carbide fuel pin was developed. This model uses thermal boundary conditions obtained from COBRA-WC output and calculates the transient thermal response of a single fuel pin to changes in internal power generation, coolant flowrate, or fuel pin physical configuration. The model uses the MITAS finite difference thermal analyzer. MITAS provides the means to input separate conductance models through the use of a user subroutine input capability. The model is a lumped-mass representation of the fuel pin using 26 nodes and 42 conductors. Run time for each transient analysis is approximately one minute of central processor time on the NOS operating system.

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## I. INTRODUCTION

The Transient Single Pin Analysis (TRANSPA) model uses the finite difference thermal analyzer code MITAS<sup>1</sup> as its basis. MITAS uses a lumped-mass thermal network and can accept user written subroutines describing physical processes of interest. In addition MITAS contains an extensive library of subroutines that includes network solutions for both transient and steady state analyses.

TRANSPA is a thermal model of a single uranium-plutonium LMFBR carbide fuel pin located within a fuel pin assembly. Its primary purpose is to provide cladding temperature histories for selected transient events such as transient

overpowers, loss of coolant simulations, or control rod transients. These cladding temperatures are used as input to the SPECKEL and ABAQUS<sup>2</sup> code that is used to assess fuel performance. Boundary conditions for TRANSPA are obtained from the results of a complete assembly analysis of the fuel pin bundle. At present these results are obtained using the COBRA-WC<sup>3</sup> thermal-hydraulics code. The procedure is to determine the steady state temperatures and flow distribution of the complete pin bundle using COBRA-WC. From these results the hot fuel pin is identified. The transient response of the hot pin or other selected pin is then analyzed using TRANSPA with flowrate and inlet temperature obtained from the COBRA-WC analysis. A description of the TRANSPA model follows.

## II. MODEL DESCRIPTION

TRANSPA is a network thermal model of a single fuel pin. Boundary conditions are coolant flowrate and inlet temperature. The heat source is uniform internal heat generation that is specified by the pin location and time in the reactor. A description of the thermal network is given followed by discussion of the special subroutines, the input power profile and the network solution characteristics.

### A. Thermal Network

The thermal network comprises 26 nodes and 42 conductors. The fuel pin is divided into five axial stations each of length 18.3 cm. Each axial segment is divided into 4 radial segments representing the fuel centerline, the fuel surface, the clad inner surface and the clad outer surface. In addition an inlet and outlet fluid node is specified for each segment. The heat generation is simulated by a source term for the fuel centerline at each segment and the heat sink is a specified coolant inlet temperature at the bottom of the fuel pin. Both radiation and conduction are used to calculate the radial heat transfer across the fuel-cladding gap. Axial conduction in the fuel and cladding is also included. The convective heat transfer coefficient is used to calculate a heat transfer effectiveness that is used to calculate thermal couplings between the coolant and the fuel cladding. The thermal network is shown in Fig. 1.

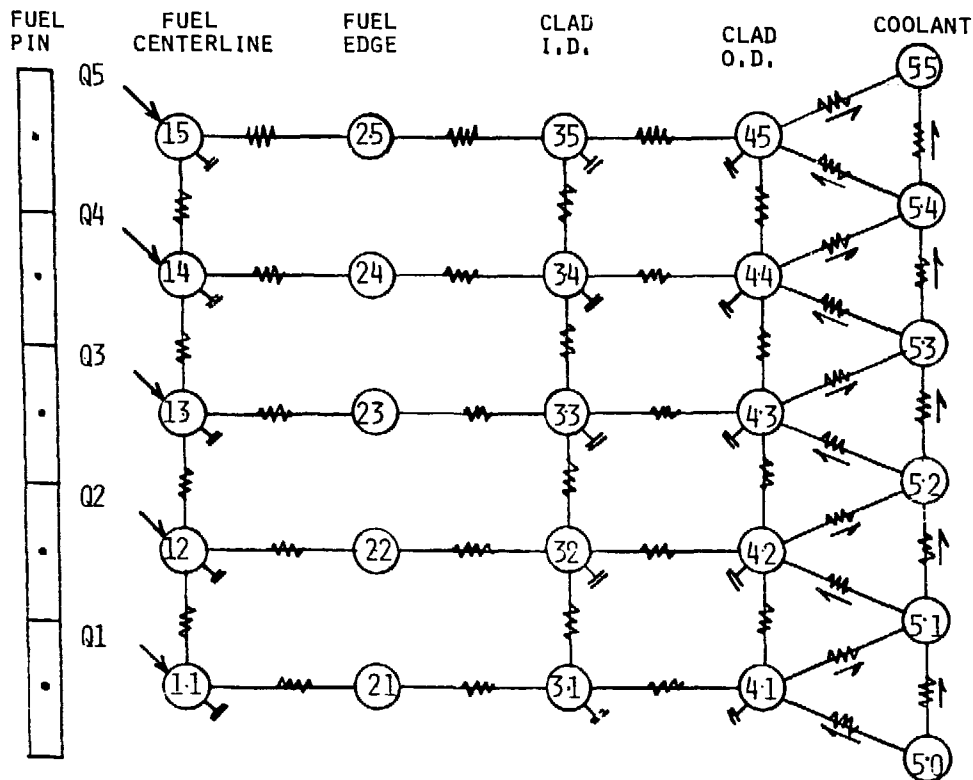


Fig. 1. Thermal network.

The thermal time constant of the model is small requiring small transient time steps. Also abrupt physical changes in the configuration lead to numerical instability in the model. These characteristics are discussed in a separate section. User subroutines required for the thermal network are discussed next.

#### B. Subroutines

User written subroutines can be easily incorporated into the MITAS code. Three subroutines are required for the TRANSPA model. The first subroutine calculates the gap conductance between the fuel and cladding. The gap conductance is a function of the interface pressure between the fuel and cladding. The subroutine uses initial conditions from either SPECKLE III or LIFE-4C code and then calculates changing interface pressure based on temperature dependent coefficients of thermal expansion and moduli of elasticity. The routine cannot currently handle plasticity. The interface pressure is then

used to approximate the gap conductance based on a curve fit generated with data predicted by the LIFE-4C code.

The second subroutine calculates the convective heat transfer coefficient of the coolant. This subroutine uses empirical data that is a function of the Reynolds and Prandtl numbers.<sup>3</sup> The convective coefficient is then used to calculate a heat exchanger effectiveness that is used to specify the thermal couplings.

The final subroutine calculates the fuel conductivity as a function of temperature and porosity.<sup>4</sup> Each of the subroutines represents a separate thermal conductance model based on either empirical data or results from a separate analysis. The structure of the MITAS code facilitates the parameterization of the TRANSPA model through the use of these user supplied subroutines.

### C. Power Profiles

The heat source term is imposed on the fuel centerline nodes of each segment. The heat source is a function of the fuel irradiation history and the pin location within the bundle and the bundle's location within the reactor. A nominal power profile is input and multiplying factors are used to define heat loads for a particular pin within the pin bundle. In addition an axial distribution of power is also imposed on the fuel pin. A transient power profile is created using a time dependent multiplying factor that is input in tabular form. A sample profile corresponding to a normal scram is shown in Fig. 2. The transient coolant flowrate is specified in a similar manner.

### D. Execution Characteristics

Because the thermal time constant is only 0.02 s the time step for the transient solution must also be small. An implicit, forward-backward differencing scheme is used for the thermal network solution. Because the solution is implicit there is no upper bound on the time step, however, if the time step greatly exceeds the thermal time constant, errors in the resulting calculation may become significant. The small thermal time constant results from large thermal couplings combined with low heat capacity within the model. The modeling of liquid metal coolants flowing at high Reynolds numbers around small diameter fuel pins with large internal heat generation leads to numerical instabilities in the solution. Especially significant are changes

## NORMAL SCRAM

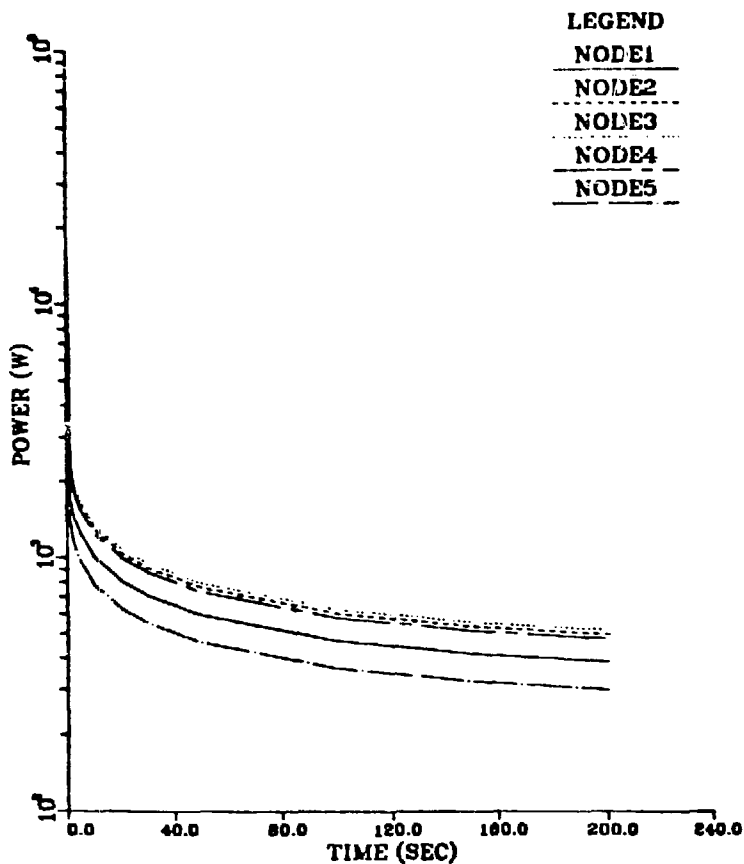


Fig. 2. Power profile for normal scram.

in the configuration that result in significant changes in the thermal couplings. Since the thermal conductance of the fuel and fuel-clad gap is strongly temperature dependent, numerical damping of these parameters must be utilized or severe temperature oscillations will occur. Because the use of these damping factors is largely empirical, sensitivity studies to determine their effect on the final results were conducted. In all cases the extent of the required damping had little influence on the final results.

This brief description of the model is intended to serve as an introduction to TRANSPA. Some of its special features and some of the difficulties encountered during its development were outlined. The following section contains a description of the output. The Appendix contains a listing of the program.

TABLE I  
TWELVE TRANSIENT EVENTS

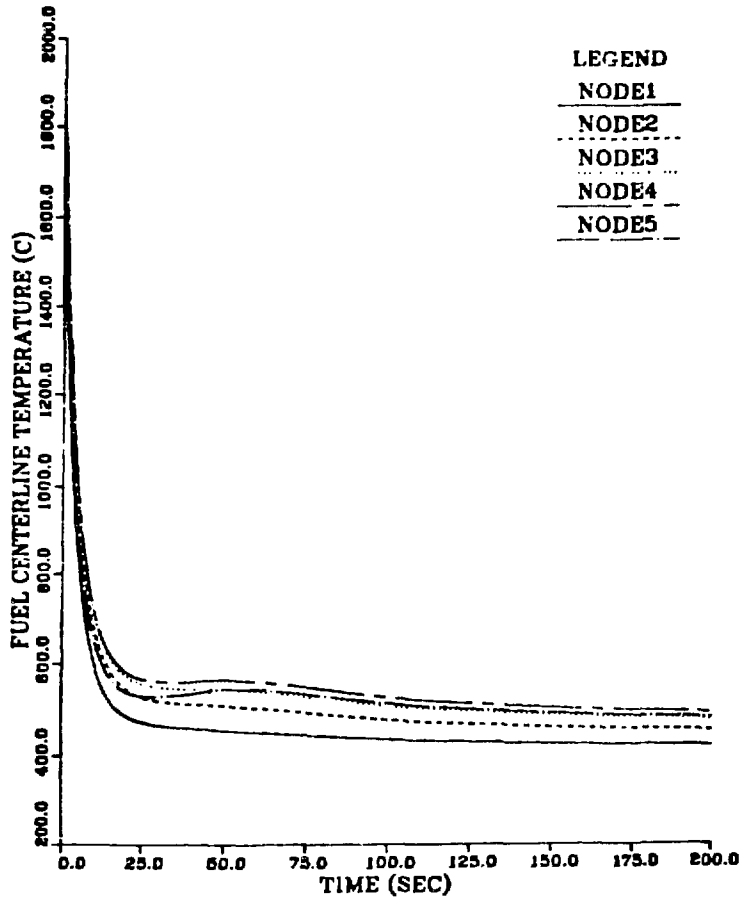
<u>Event</u>	<u>Initial Condition</u>
1. Normal scram	Full power
2. Rod drop	Full power
3. 0.5 $\phi$ /s, 100% flow to 460 MW, primary trip	Full power
4. 3.4 $\phi$ /s, 100% flow to 486 MW, primary trip	Full power
5. Continuous flow reduction, primary trip	93% power and flow
6. 0.5 $\phi$ /s, 100% flow to 500 MW, secondary trip	Full power
7. 3.4 $\phi$ /s, 100% flow to 526 MW, secondary trip	Full power
8. 25 $\phi$ /s, 100% flow to 500 MW, primary trip	Full power
9. 25 $\phi$ /s, 100% flow to 528 MW, secondary trip	Full power
10. Total loss electric power	Full power
11. Continuous flow reduction, secondary trip	93% power and flow
12. Primary pump seizure	Full power

### III. OUTPUT AND RESULTS

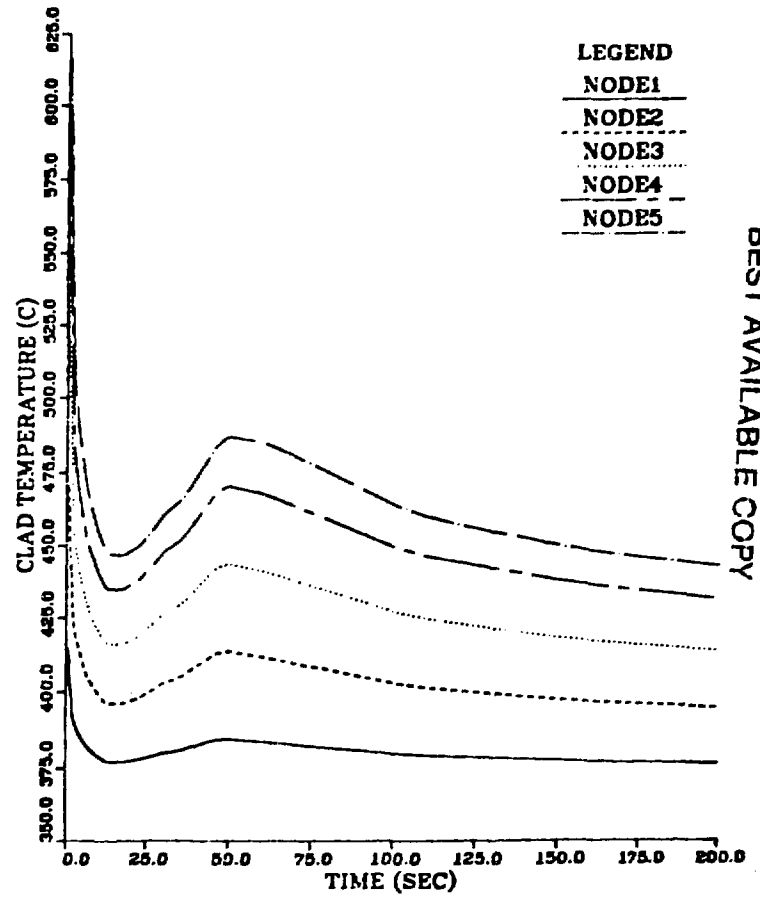
TRANSPA is used to evaluate the transient thermal response of a single fuel pin to the twelve events shown in Table I. For each event a transient power profile and coolant flowrate are specified. For each case a steady state solution is obtained that serves as the initial condition for the transient. The model then calculates the transient response of the fuel centerline, the outside clad, and the fluid outlet temperatures. These temperatures are saved on a separate file and plotted using a separate program. Examples of the output are shown in Figs. 3, 4, and 5. Each figure contains five curves corresponding to each of the five axial nodes in TRANSPA.

The model is set up to analyze all of the twelve transient events in a single run. Logic to accomplish this is contained in the MITAS execution

### NORMAL SCRAM



### NORMAL SCRAM



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Fig. 3. Fuel centerline temperature for normal scram.

Fig. 4. Clad temperature for normal scram.



## NORMAL SCRAM

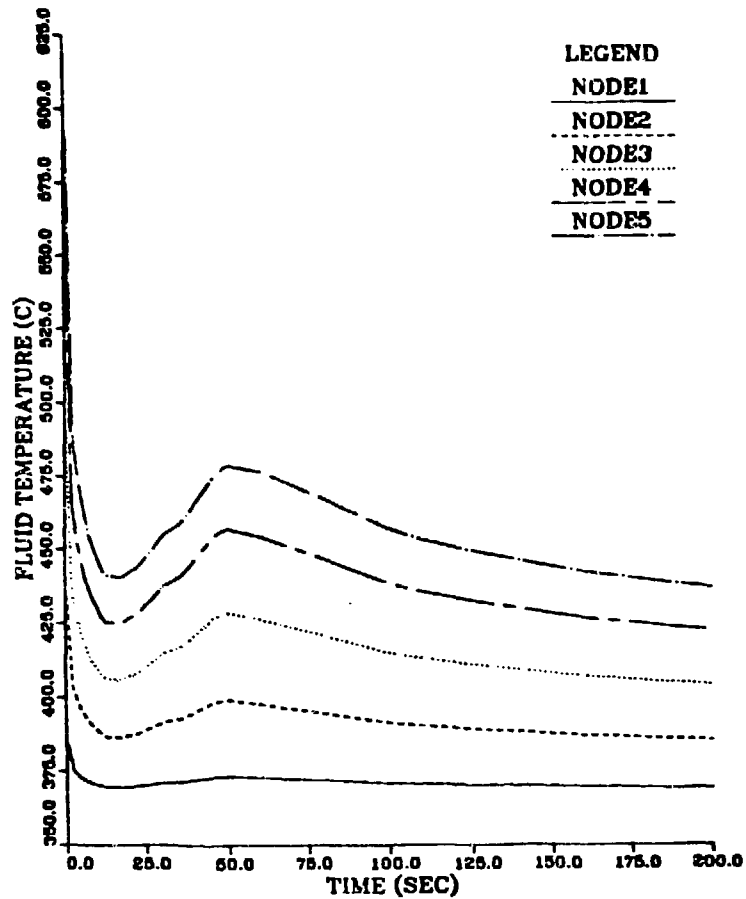


Fig. 5. Coolant temperature for normal scram.

routine. In addition two steady state at 93% power and coolant flow rate conditions are analyzed. The solution time step is varied during each calculation depending on the rate of change of the power profile or coolant flowrate. Execution time for the twelve transient cases requires approximately ten minutes of central processing time on the NOS system.

### V. REFERENCES

1. G. M. Holmstead, R. E. Kannady, Jr., R. J. Connor, and E. Shirley, "Martin Marietta Interactive Thermal Analysis System, MITAS-II," User's Manual, May 1976.

2. ABAQUS Computer Program Manuals (Hibbitt, Karlsson & Sorensen, Inc., Providence, Rhode Island, 1981), Vol. 1-4.
3. E. P. Coomes, E. V. Khan, T. C. George, R. L. Cheatham, and D. R. Rector, "User's Manual, COBRA-WC: A Version of COBRA for Single Phase and Single and Multiassembly Thermal-Hydraulic Transient Analysis," Pacific Northwest Laboratories report PNL-3259, 1979.
4. T. L. George and K. L. Peddicord, "SPECKLE III: A Computer Code for Calculating the Behavior of Sphere Pac Fuel Pins," Oregon State University Rept. OSU-EIR-64, December 1982.

APPENDIX

MITAS II INPUT DATA FOR TRANSPA MODEL

START OF INPUT DECK PROCESSING

INPUT CARD COL. =	1	2	3	4	5	6	7	8	EDIT NO.	CARD NO.
BCD 3	TITLE DATA								A. 1	1
BCD 9	TRANSPA: TRANSIENT SINGLE PIN ANALYSIS								A. 2	2
BCD 9	CARBIDE FUELS PROGRAM								A. 3	3
BCD 9	LANL MAY 1984								A. 4	4
BCD 95	AXIAL NODES W/ AXIAL CONDUCTION								A. 5	5
BCD 9	AC3 CONFIGURATION, PIN 10 (MAX TEMP)								A. 6	6
BCD 9	THERMAL CONDUCTIVITY IS CURVEFIT FR SPECKLE POROSITY=0.23								A. 7	7
BCD 9	INCLUDES GAP CONDUCTANCE MODEL								A. 8	8
BCD 9	UNITS ARE G-CM-S-DEG C-W								A. 9	9
END									A. 10	10

TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO.

CARD NO.

BCD 3NODE DATA				A.		
GEN	11, 5, 1, 1500.0, 1.0\$	FUEL CENTERLINE		A.	11	11
++++	11,	.15000E+04.	.10000E+01	A.	12	12
++++	12,	.15000E+04.	.10000E+01			
++++	13,	.15000E+04.	.10000E+01			
++++	14,	.15000E+04.	.10000E+01			
++++	15,	.15000E+04.	.10000E+01			
GEN	21, 5, 1, 593.0, -1.\$	FUEL OD		A.	13	13
++++	21,	.59300E+03.	-.10000E+01			
++++	22,	.59300E+03.	-.10000E+01			
++++	23,	.59300E+03.	-.10000E+01			
++++	24,	.59300E+03.	-.10000E+01			
++++	25,	.59300E+03.	-.10000E+01			
GEN	31, 5, 1, 593.0, 1.0\$	CLAD ID		A.	14	14
++++	31,	.59300E+03.	.10000E+01			
++++	32,	.59300E+03.	.10000E+01			
++++	33,	.59300E+03.	.10000E+01			
++++	34,	.59300E+03.	.10000E+01			
++++	35,	.59300E+03.	.10000E+01			
GEN	41, 5, 1, 593.0, 1.0\$	CLAD OD		A.	15	15
++++	41,	.59300E+03.	.10000E+01			
++++	42,	.59300E+03.	.10000E+01			
++++	43,	.59300E+03.	.10000E+01			
++++	44,	.59300E+03.	.10000E+01			
++++	45,	.59300E+03.	.10000E+01			
GEN	51, 5, 1, 375.0, -1.\$	FLUID(COOLANT)		A.	16	16
++++	51,	.37500E+03.	-.10000E+01			
++++	52,	.37500E+03.	-.10000E+01			
++++	53,	.37500E+03.	-.10000E+01			
++++	54,	.37500E+03.	-.10000E+01			
++++	55,	.37500E+03.	-.10000E+01			
	-50, 362.0, 0.0\$	COOLANT INLET		A.	17	17
	END			A.	18	18

RFL TO 116127B REQUESTED. OLD FL=200000B ,NEW FL=200000B.

NODE SUMMARY

THERE ARE 15 DIFFUSION NODES, 10 ARITHMETIC NODES, 0 HEATER NODES, AND 1 BOUNDARY NODES FOR A TOTAL OF 26 NODES IN THE NETWORK

NODE DIRECTORY  
INPUT/INTERNAL

DIFFUSION NODES

11=	1 \$	12=	2 \$	13=	3 \$	14=	4 \$	15=	5 \$	31=	6 \$	32=	7 \$	33=	8 \$	34=	9
35=	10 \$	41=	11 \$	42=	12 \$	43=	13 \$	44=	14 \$	45=	15 \$						

ARITHMETIC NODES

21=	16 \$	22=	17 \$	23=	18 \$	24=	19 \$	25=	20 \$	51=	21 \$	52=	22 \$	53=	23 \$	54=	24
55=	25 \$																

HEATER NODES

++NONE++

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO.

CARD NO.

BCD 3CONDUCTOR DATA	A.	19	19
RFL TO 000008 REQUESTED. OLD FL=200008 ,NEW FL=200008.			
GEN 41,5,1,-50,1,41,1, 1.0% FLUID COUPLING	A.	20	20
+++++ 41, -50, 41, .10000E+01			
+++++ 42, -51, 42, .10000E+01			
+++++ 43, -52, 43, .10000E+01			
+++++ 44, -53, 44, .10000E+01			
+++++ 45, -54, 45, .10000E+01			
GEN 51,5,1,-50,1,51,1, 1.0	A.	21	21
+++++ 51, -50, 51, .10000E+01			
+++++ 52, -51, 52, .10000E+01			
+++++ 53, -52, 53, .10000E+01			
+++++ 54, -53, 54, .10000E+01			
+++++ 55, -54, 55, .10000E+01			
GEN 61,5,1,-41,1,51,1, 1.0	A.	22	22
+++++ 61, -41, 51, .10000E+01			
+++++ 62, -42, 52, .10000E+01			
+++++ 63, -43, 53, .10000E+01			
+++++ 64, -44, 54, .10000E+01			
+++++ 65, -45, 55, .10000E+01			
GEN 11,5,1,11,1,21,1, 1.0% RADIAL CONDUCTION	A.	23	23
+++++ 11, 11, 21, .10000E+01			
+++++ 12, 12, 22, .10000E+01			
+++++ 13, 13, 23, .10000E+01			
+++++ 14, 14, 24, .10000E+01			
+++++ 15, 15, 25, .10000E+01			
GEN 21,5,1,21,1,31,1, 1.0	A.	24	24
+++++ 21, 21, 31, .10000E+01			
+++++ 22, 22, 32, .10000E+01			
+++++ 23, 23, 33, .10000E+01			
+++++ 24, 24, 34, .10000E+01			
+++++ 25, 25, 35, .10000E+01			
GEN 31,5,1,31,1,41,1, 1.0	A.	25	25
+++++ 31, 31, 41, .10000E+01			
+++++ 32, 32, 42, .10000E+01			
+++++ 33, 33, 43, .10000E+01			
+++++ 34, 34, 44, .10000E+01			
+++++ 35, 35, 45, .10000E+01			
GEN 111,4,1,11,1,12,1, 1.0% AXIAL CONDUCTION	A.	26	26
+++++ 111, 11, 12, .10000E+01			
+++++ 112, 12, 13, .10000E+01			
+++++ 113, 13, 14, .10000E+01			
+++++ 114, 14, 15, .10000E+01			
GEN 131,4,1,31,1,32,1, 1.0	A.	27	27
+++++ 131, 31, 32, .10000E+01			
+++++ 132, 32, 33, .10000E+01			
+++++ 133, 33, 34, .10000E+01			
+++++ 134, 34, 35, .10000E+01			
GEN 141,4,1,41,1,42,1, 1.0	A.	28	28
+++++ 141, 41, 42, .10000E+01			
+++++ 142, 42, 43, .10000E+01			
+++++ 143, 43, 44, .10000E+01			
+++++ 144, 44, 45, .10000E+01			

TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO.

CARD NO.

END

A. 29

29

RFL TO 116135B REQUESTED. OLD FL=200000B ,NEW FL=200000B.

CONDUCTOR SUMMARY

THERE ARE 42 LINEAR CONDUCTORS AND 0 NON-LINEAR CONDUCTORS , FOR A TOTAL OF 42 CONDUCTORS IN THIS NETWORK

CONDUCTOR DIRECTORY  
INPUT=INTERNAL

41=	1 \$	42=	2 \$	43=	3 \$	44=	4 \$	45=	5 \$	51=	6 \$	52=	7 \$	53=	8 \$	54=	9
55=	10 \$	61=	11 \$	62=	12 \$	63=	13 \$	64=	14 \$	65=	15 \$	11=	16 \$	12=	17 \$	13=	18
14=	19 \$	15=	20 \$	21=	21 \$	22=	22 \$	23=	23 \$	24=	24 \$	25=	25 \$	31=	26 \$	32=	27
33=	28 \$	34=	29 \$	35=	30 \$	111=	31 \$	112=	32 \$	113=	33 \$	114=	34 \$	131=	35 \$	132=	36
133=	37 \$	134=	38 \$	141=	39 \$	142=	40 \$	143=	41 \$	144=	42 \$						

RFL TO 116150B REQUESTED. OLD FL=200000B ,NEW FL=200000B.

RFL TO 116043B REQUESTED. OLD FL=200000B ,NEW FL=200000B.

INPUT CARD COL.12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT ND.

CARD NO.

BCD 3CONSTANTS DATA	A.	30	30
1.0.23 \$ POROSITY OF FUEL	A.	31	31
2.0.24 \$ CLAD CONDUCTIVITY - AXIAL ONLY (W/CM-C)	A.	32	32
3.13.51 \$ DENSITY FUEL (G/CM**3)	A.	33	33
4.7.64 \$ DENSITY CLAD (G/CM**3)	A.	34	34
5.0.250 \$ CP FUEL (W-S/G-K)	A.	35	35
6.0.505 \$ CP CLAD (W-S/G-K)	A.	36	36
7.1.04 \$ RADIAL POWER FACTOR	A.	37	37
8.14.0E6\$ INITIAL INTERFACE PRESSURE	A.	38	38
9.228.4 \$ COOLANT FLOWRATE (G/S)	A.	39	39
11.0.838 \$ FUEL OD (CM)	A.	40	40
12.0.838 \$ CLAD ID (CM)	A.	41	41
13.0.940 \$ CLAD OD (CM)	A.	42	42
14.18.27 \$ NODE LENGTH (CM)	A.	43	43
15.1321.6\$ NOMINAL POWER DENSITY (W/CM**3)	A.	44	44
16.1.26 \$ CP COOLANT (W-S/G-K)	A.	45	45
17.0.536 \$ HYDRAULIC DIA OF CHANNEL	A.	46	46
18.0.1974\$ FLOW AREA OF CHANNEL	A.	47	47
21.0.0 \$ FUEL VOLUME (CM**3)	A.	48	48
22.0.0 \$ CLAD VOLUME (CM**3)	A.	49	49
23.0.0 \$ FUEL AXIAL CONDUCTANCE (W/K)	A.	50	50
24.0.0 \$ 0.5*CLAD AXIAL CONDUCTANCE (W/K)	A.	51	51
25.0.0 \$ FUEL RADIAL CONDUCTANCE (W/K)	A.	52	52
26.0.0 \$ CLAD RADIAL CONDUCTANCE (W/K)	A.	53	53
27.0.0 \$ CAPACITANCE OF FUEL (W-S/K)	A.	54	54
28.0.0 \$ CAPACITANCE OF CLAD (W-S/K)	A.	55	55
29.0.0 \$ FUEL SURFACE AREA (CM**2)	A.	56	56
30.0.0 \$ TRANSIENT HEAT FLUX NORMALIZED (W/CM**3)	A.	57	57
31.0.0 \$ TRANSIENT FLOWRATE (G/S)	A.	58	58
32.0.0 \$ FLOWRATE*CP COOLANT	A.	59	59
41.0.0\$ CLAD-FUEL INTERFACE PRESSURE NODE 1	A.	60	60
42.0.0\$ CLAD-FUEL INTERFACE PRESSURE NODE 2	A.	61	61
43.0.0\$ CLAD-FUEL INTERFACE PRESSURE NODE 3	A.	62	62
44.0.0\$ CLAD-FUEL INTERFACE PRESSURE NODE 4	A.	63	63
45.0.0\$ CLAD-FUEL INTERFACE PRESSURE NODE 5	A.	64	64
51.0.903 \$ RELATIVE POWER NODE 1	A.	65	65
52.1.159 \$ RELATIVE PDWER NODE 2	A.	66	66
53.1.203 \$ RELATIVE POWER NODE 3	A.	67	67
54.1.114 \$ RELATIVE POWER NODE 4	A.	68	68
55.0.7C2 \$ RELATIVE POWER NODE 5	A.	69	69
60.0.48Q2 \$ 0 TH COEFFICIENT FUEL CONDUCTIVITY (W/CM-K)	A.	70	70
61.-1.367E-3 \$ 1 ST COEFFICIENT FUEL CONDUCTIVITY (W/CM-K)	A.	71	71
62.1.607E-6 \$ 2 ND COEFFICIENT (T**2)	A.	72	72
63.-8.379E-10\$ 3 RD COEFFICIENT (T**3)	A.	73	73
64.1.643E-13 \$ 4 TH COEFFICIENT (T**4)	A.	74	74
68.0.115 \$ 0 TH COEFFICIENT CLAD CONDUCTIVITY (W/CM-C)	A.	75	75
69.1.30E-4 \$ 1 ST COEFFICIENT (T)	A.	76	76
71.0.0\$ FUEL-CLADDING GAP CONDUCTANCE	A.	77	77
72.0.0\$	A.	78	78
73.0.0\$	A.	79	79
74.0.0\$	A.	80	80
75.0.0\$	A.	81	81



TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8	EDIT NO.	CARD NO.
90,0.95 \$ GAP CONDUCTANCE DAMPING FACTOR	A.	82
92,CGAP\$ LABEL	A.	83
93,IPRESS\$ LABEL	A.	84
94,0.750 \$ FUEL TEMPERATURE WEIGHT FACTOR FOR K	A.	85
95,0.70 \$ FUEL CONDUCTANCE DAMPING FACTOR	A.	86
96,5.669E-12\$ STEPHAN BOLTZMANN CONSTANT	A.	87
97,CNST\$ LABEL	A.	88
98,EFF \$ LABEL	A.	89
99,3.14159\$	A.	90
111,1500.0\$ SS FUEL C/L TEMP	A.	91
112,1500.0\$	A.	92
113,1500.0\$	A.	93
114,1500.0\$	A.	94
115,1500.0\$	A.	95
121,593.0\$ SS FUEL EDGE TEMP	A.	96
122,593.0\$	A.	97
123,593.0\$	A.	98
124,593.0\$	A.	99
125,593.0\$	A.	100
131,593.0\$ SS CLAD OD TEMP	A.	101
132,593.0\$	A.	102
133,593.0\$	A.	103
134,593.0\$	A.	104
135,593.0\$	A.	105
141,593.0\$ SS CLAD OD TEMP	A.	106
142,593.0\$	A.	107
143,593.0\$	A.	108
144,593.0\$	A.	109
145,593.0\$	A.	110
155,375.0\$ SS COOLANT OUTLET TEMP	A.	111
ITEST,O \$ TRANSIENT FLAG (1=TRANSIENT)	A.	112
NTEST,O \$ NUMBER OF TIME STEPS IN TRANSIENT	A.	113
SBCNST = 1.0\$	A.	114
ABSZRO = -273.0\$	A.	115
DRLXCA = 0.005\$	A.	116
ARLXCA = 0.005\$	A.	117
EBALSA = 500.0\$	A.	118
EBALNA = 100.0\$	A.	119
NDSTOR = 200\$	A.	120
ITERMX = 200\$	A.	121
EXTLIM = 100.0\$	A.	122
ITERXT = 5\$	A.	123
END	A.	124

RFL TO 116374B REQUESTED. OLD FL=200000B ,NEW FL=200000B.

CONSTANTS DIRECTORY

INPUT=INTERNAL																	
1=	1 \$	2=	2 \$	3=	3 \$	4=	4 \$	5=	5 \$	6=	6 \$	7=	7 \$	8=	8 \$	9=	9
11=	10 \$	12=	11 \$	13=	12 \$	14=	13 \$	15=	14 \$	16=	15 \$	17=	16 \$	18=	17 \$	21=	18
22=	19 \$	23=	20 \$	24=	21 \$	25=	22 \$	26=	23 \$	27=	24 \$	28=	25 \$	29=	26 \$	30=	27
31=	28 \$	32=	29 \$	41=	30 \$	42=	31 \$	43=	32 \$	44=	33 \$	45=	34 \$	51=	35 \$	52=	36

TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

CARBIDE FUELS PROGRAM

LANL MAY 198

53=	37 \$	54=	38 \$	55=	39 \$	60=	40 \$	61=	41 \$	62=	42 \$	63=	43 \$	64=	44 \$	68=	45
69=	46 \$	71=	47 \$	72=	48 \$	73=	49 \$	74=	50 \$	75=	51 \$	90=	52 \$	92=	53 \$	93=	54
94=	55 \$	95=	56 \$	96=	57 \$	97=	58 \$	98=	59 \$	99=	60 \$	111=	61 \$	112=	62 \$	113=	63
114=	64 \$	115=	65 \$	121=	66 \$	122=	67 \$	123=	68 \$	124=	69 \$	125=	70 \$	131=	71 \$	132=	72
133=	73 \$	134=	74 \$	135=	75 \$	141=	76 \$	142=	77 \$	143=	78 \$	144=	79 \$	145=	80 \$	155=	81

TRANSPA: TRANSIENT SINGLE PIN ANALYSIS CARBIDE FJJE-S PROGRAM INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO. CARD NO.

BCD	JARRAY DATA	A.	125
2\$	TIMEND ARRAY FOR TRANSIENTS	A.	126
200.0.	0.200.0.50.0.12.0.600.0.50.0.12.0.12.0.12.0	A.	127
200.0.	0.600.0.200.0.END\$	A.	128
8.	SPACE,5.END\$ COMPUTED HEAT GENERATION EACH NODE	A.	129
40.	SPACE,150.END\$	A.	130
41.	SPACE,150.END\$	A.	131
42.	SPACE,150.END\$	A.	132
43.	SPACE,150.END\$	A.	133
44.	SPACE,150.END\$	A.	134
45.	SPACE,150.END\$	A.	135
51.	SPACE,150.END\$	A.	136
52.	SPACE,150.END\$	A.	137
53.	SPACE,150.END\$	A.	138
54.	SPACE,150.END\$	A.	139
55.	SPACE,150.END\$	A.	140
61.	SPACE,150.END\$	A.	141
62.	SPACE,150.END\$	A.	142
63.	SPACE,150.END\$	A.	143
64.	SPACE,150.END\$	A.	144
65.	SPACE,150.END\$	A.	145
66.	SPACE,150.END\$	A.	146
71.	SPACE,150.END\$	A.	147
72.	SPACE,150.END\$	A.	148
73.	SPACE,150.END\$	A.	149
74.	SPACE,150.END\$	A.	150
75.	SPACE,150.END\$	A.	151
101\$	NORMAL SCRAM - POWER	A.	152
0.0.	1.0000,	A.	153
3.0.	0.1120,	A.	154
10.1.	0.0800,	A.	155
50.1.	0.0474,	A.	156
200.1.	0.0310, END\$	A.	157
201\$	NORMAL SCRAM - FLOW	A.	158
0.000.	1.0000,	A.	159
3.016.	0.6778,	A.	160
6.010.	0.4938,	A.	161
8.999.	0.3863,	A.	162
19.998.	0.2084,	A.	163
35.110.	0.1293,	A.	164
50.071.	.09294,	A.	165
74.952.	.09255,	A.	166
200.029.	.09546, END\$	A.	167
102\$	RDD DROP - POWER	A.	168
0.0.	1.0000,	A.	169
19.3.	0.1400,	A.	170
22.3.	0.1050,	A.	171
50.4.	0.0553,	A.	172
198.5.	0.0317, END\$	A.	173
202\$	RDD DROP - FLOW	A.	174
0.000.	1.0000,	A.	175
18.300.	1.0000,	A.	176

0.1400,	2.0,	0.1220
0.1050,	5.0,	0.0989
0.0645,	30.0,	0.0566
0.0374,	150.1,	0.0333
0.8923,	2.000,	0.7713
0.6015,	5.053,	0.5413
0.4511,	8.069,	0.4146
0.3596,	15.047,	0.2648
0.1695,	30.059,	0.1425
0.1123,	45.099,	.09833
.09269,	60.026,	.09275
.09308,	150.103,	.09434
0.7620,	18.5,	0.7590
0.1220,	21.3,	0.1120
0.0989,	30.4,	0.0755
0.0399,	148.4,	0.0346
0.8923,	20.300,	0.7713

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO. CARD NO.

21.316	0.6778	22.347	0.6015	23.353	0.5413	A.	177
24.310	0.4938	25.336	0.4511	26.369	0.4146	A.	178
27.299	0.3863	28.298	0.3596	33.347	0.2648	A.	179
38.298	0.2084	43.395	0.1695	48.359	0.1425	A.	180
53.410	0.1293	58.325	0.1123	63.399	.09833	A.	181
68.371	.09294	73.349	.09269	78.326	.09275	A.	182
93.252	.09255	118.291	.09308	168.403	.09434	A.	183
218.329	.09546	END\$				A.	184
103\$ 0.5C/S	PRIMARY TRIP	5.0	1.0250	10.0	1.0512	A.	185
0.0	1.0000	20.0	1.1105	26.0	1.1500	A.	186
15.0	1.0798	30.0	0.0960	35.0	0.0912	A.	187
27.0	0.1378	41.0	0.0850	45.0	0.0828	A.	188
40.3	0.0855	END\$				A.	189
50.0	0.0822	END\$				A.	190
203 0.0.1.0.50.0.1.0	END\$					A.	191
104\$ 3.4C/S	PRIMARY TRIP	1.0	1.0355	2.0	1.0722	A.	192
0.0	1.0000	4.0	1.1495	5.0	1.1898	A.	193
3.0	1.1105	5.61	0.5245	5.68	0.3932	A.	194
5.6	1.2150	7.0	0.1290	7.8	0.1200	A.	195
6.0	0.1812	8.4	0.1150	9.0	0.1105	A.	196
8.0	0.1182	12.0	0.0968	END\$		A.	197
10.0	0.1050	END\$				A.	198
204 0.0.1.0.12.0.1.0	END\$					A.	199
105\$ CONTINUOUS FLOW	REDUCTION	1.02	0.9988	1.98	0.9923	A.	200
0.00	1.0000	4.08	0.9676	5.04	0.9534	A.	201
3.00	0.9816	7.02	0.9255	8.04	0.9123	A.	202
6.06	0.9392	10.02	0.8877	11.10	0.8752	A.	203
9.00	0.8997	13.08	0.8538	13.98	0.1805	A.	204
12.06	0.8643	16.02	0.0954	16.98	0.0901	A.	205
15.00	0.1037	19.08	0.0822	20.04	0.0793	A.	206
18.06	0.0856	22.02	0.0747	23.04	0.0728	A.	207
21.06	0.0768	25.02	0.0694	25.98	0.0679	A.	208
24.00	0.0712	28.08	0.0651	29.10	0.0639	A.	209
27.06	0.0663	31.08	0.0619	32.04	0.0609	A.	210
30.06	0.0629	34.02	0.0593	35.04	0.0585	A.	211
33.06	0.0601	37.02	0.0571	38.04	0.0565	A.	212
36.00	0.0578	40.02	0.0553	41.10	0.0547	A.	213
39.00	0.0559	43.08	0.0536	44.10	0.0531	A.	214
42.06	0.0541	46.08	0.0522	47.04	0.0518	A.	215
45.06	0.0527	49.02	0.0509	50.04	0.0505	A.	216
48.06	0.0513	52.02	0.0495	53.04	0.0494	A.	217
51.06	0.0502	55.02	0.0488	55.98	0.0484	A.	218
54.00	0.0491	58.08	0.0478	59.10	0.0475	A.	219
57.00	0.0481	75.06	0.0437	90.00	0.0411	A.	220
60.06	0.0472	120.06	0.0378	135.00	0.0366	A.	221
105.00	0.0393	165.06	0.0347	180.00	0.0340	A.	222
150.06	0.0356	210.06	0.0328	225.00	0.0323	A.	223
195.00	0.0334	255.00	0.0314	270.06	0.0310	A.	224
240.06	0.0318	300.06	0.0302	315.06	0.0299	A.	225
285.06	0.0306	345.06	0.0293	360.06	0.0291	A.	226
330.06	0.0296	390.00	0.0286	405.06	0.0283	A.	227
375.00	0.0288					A.	228

TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO.

CARD NO.

420.12	0.0281	435.06	0.0279	450.06	0.0277	A.	229
465.00	0.0275	480.06	0.0273	495.06	0.0271	A.	230
510.00	0.0270	525.06	0.0268	540.00	0.0266	A.	231
555.12	0.0265	570.06	0.0263	585.06	0.0262	A.	232
600.00	0.0260	END\$				A.	233
205\$ CONTINUOUS FLOW		REDUCTION,		PRIMARY TRIP - FLOW		A.	234
0.00	1.0000	1.02	0.9818	1.98	0.9416	A.	235
3.00	0.8994	4.08	0.8577	5.04	0.8213	A.	236
6.06	0.7883	7.02	0.7584	8.04	0.7312	A.	237
9.00	0.7055	10.02	0.6839	11.10	0.6612	A.	238
12.06	0.6423	13.08	0.6242	13.98	0.6023	A.	239
15.00	0.5489	16.02	0.4993	16.98	0.4585	A.	240
18.06	0.4188	19.08	0.3868	20.04	0.3587	A.	241
21.06	0.3340	22.02	0.3121	23.04	0.2926	A.	242
24.00	0.2751	25.02	0.2594	25.98	0.2452	A.	243
27.06	0.2311	28.08	0.2195	29.10	0.2085	A.	244
30.06	0.1991	31.08	0.1901	32.04	0.1818	A.	245
33.06	0.1742	34.02	0.1671	35.04	0.1605	A.	246
36.00	0.1543	37.02	0.1485	38.04	0.1430	A.	247
39.00	0.1378	40.02	0.1328	41.10	0.1276	A.	248
42.06	0.1232	43.08	0.1190	44.10	0.1151	A.	249
45.06	0.1113	46.08	0.1078	47.04	0.1055	A.	250
48.06	0.1043	49.02	0.1037	50.04	0.1033	A.	251
51.06	0.1030	52.02	0.1029	53.04	0.1027	A.	252
54.00	0.1027	55.02	0.1026	90.00	0.1029	A.	253
60.06	0.1026	75.06	0.1033	135.00	0.1040	A.	254
105.00	0.1031	120.06	0.1034	180.00	0.1050	A.	255
150.06	0.1045	165.06	0.1048	225.00	0.1063	A.	256
195.00	0.1053	210.06	0.1058	270.06	0.1080	A.	257
240.06	0.1070	255.00	0.1075	315.06	0.1089	A.	258
285.06	0.1084	300.06	0.1087	360.06	0.1096	A.	259
330.06	0.1091	345.06	0.1093	405.06	0.1100	A.	260
375.00	0.1097	390.00	0.1099	450.06	0.1102	A.	261
420.12	0.1101	435.06	0.1101			A.	262
540.00	0.1102					A.	263
555.12	0.1101	570.06	0.1101	585.06	0.1100	A.	264
600.00	0.1099	END\$				A.	265
106\$ O.5C/S.		SECONDARY TRIP				A.	266
0.0	1.0000	5.0	1.0250	10.0	1.0512	A.	267
15.0	1.0798	20.0	1.1105	26.0	1.1500	A.	268
27.0	1.1560	30.0	1.1770	35.0	1.2140	A.	269
40.3	1.2500	41.0	0.1465	45.0	0.0970	A.	270
50.0	0.0908	END\$				A.	271
206\$ O.O.1.O.50.O.1.O.END\$						A.	272
107\$ 3.AC/S.		SECONDARY TRIP				A.	273
0.0	1.0000	1.0	1.0355	2.0	1.0722	A.	274
3.0	1.1105	4.0	1.1495	5.0	1.1898	A.	275
5.6	1.2150	6.0	1.2322	7.0	1.2768	A.	276
7.3	1.3138	7.816	0.8838	7.962	0.7690	A.	277
8.0	0.6898	8.4	0.2708	9.0	0.2062	A.	278
10.0	0.1845	12.0	0.1595	END\$		A.	279
207\$ O.O.1.O.12.O.1.O.END\$						A.	280

TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO.

CARD NO.

111\$ CONTINUOUS FLOW REDUCTION.	SECONDARY TRIP	POWER	211\$ CONTINUOUS FLOW REDUCTION.	SECONDARY TRIP	FLOW	281	281
0.00	1.0000	1.02	0.9997	1.98	0.9974	A.	281
3.00	0.9932	4.08	0.9869	4.98	0.9803	A.	282
6.00	0.9722	7.02	0.9636	8.04	0.9547	A.	283
9.06	0.9458	10.04	0.9378	11.04	0.9290	A.	284
12.06	0.9204	13.08	0.9117	14.10	0.9031	A.	285
15.00	0.8953	16.02	0.8868	17.04	0.8783	A.	286
18.06	0.8700	19.08	0.8621	19.98	0.1190	A.	287
21.00	0.0591	22.02	0.0921	23.10	C.0872	A.	288
24.00	0.0837	25.02	0.0804	26.04	0.0776	A.	289
27.06	0.0753	28.08	0.0732	29.04	0.0716	A.	290
30.06	0.0699	31.08	0.0682	32.10	0.0667	A.	291
33.00	0.0655	34.02	0.0642	35.04	0.0631	A.	292
36.06	0.0620	37.02	0.0612	38.04	0.0603	A.	293
39.06	0.0594	40.08	0.0586	41.04	0.0579	A.	294
42.06	0.0572	43.08	0.0565	44.10	0.0559	A.	295
45.00	0.0553	46.02	0.0548	47.04	0.0542	A.	296
48.12	0.0537	49.02	0.0532	50.04	0.0527	A.	297
51.06	0.0523	52.02	0.0518	53.04	0.0514	A.	298
54.06	0.0510	55.08	0.0506	55.98	0.0502	A.	299
57.00	0.0498	58.08	0.0495	59.10	0.0491	A.	300
60.00	0.0488	75.00	0.0448	90.00	0.0419	A.	301
105.12	0.0398	120.12	0.0382	135.00	0.0370	A.	302
150.06	0.0359	165.00	0.0350	180.06	0.0342	A.	303
195.06	0.0336	210.06	0.0330	225.06	0.0324	A.	304
240.06	0.0320	255.06	0.0315	270.00	0.0311	A.	305
285.06	0.0307	300.06	0.0304	315.06	0.0300	A.	306
330.06	0.0297	345.06	0.0294	360.06	0.0292	A.	307
375.06	0.0289	390.06	0.0286	405.06	0.0284	A.	308
420.00	0.0282	435.00	0.0280	450.00	0.0278	A.	309
465.12	0.0276	480.12	0.0274	495.00	0.0272	A.	310
510.00	0.0270	525.00	0.0268	540.00	0.0267	A.	311
555.06	0.0265	570.06	0.0264	585.06	0.0262	A.	312
600.00	0.0261	END\$				A.	313
		END\$				A.	314
		END\$				A.	315
		END\$				A.	316
		END\$				A.	317
		END\$				A.	318
		END\$				A.	319
		END\$				A.	320
		END\$				A.	321
		END\$				A.	322
		END\$				A.	323
		END\$				A.	324
		END\$				A.	325
		END\$				A.	326
		END\$				A.	327
		END\$				A.	328
		END\$				A.	329
		END\$				A.	330
		END\$				A.	331
		END\$				A.	332

## TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

## CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL.	1	2	3	4	5	6	7	8	EDIT NO.	CARD NO.
45.00,	0.1382,	46.02,	0.1322,	47.04,	0.1286	A.	333	333		
48.12,	0.1241,	49.02,	0.1206,	50.04,	0.1164	A.	334	334		
51.06,	0.1138,	52.02,	0.1125,	53.04,	0.1117	A.	335	335		
54.06,	0.1113,	55.08,	0.1109,	55.98,	0.1107	A.	336	336		
57.00,	0.1106,	58.08,	0.1105,	59.10,	0.1104	A.	337	337		
60.00,	0.1103,	75.00,	0.1108,	90.00,	0.1107	A.	338	338		
105.12,	0.1106,	120.12,	0.1107,	135.00,	0.1110	A.	339	339		
150.06,	0.1116,	165.00,	0.1120,	180.06,	0.1124	A.	340	340		
195.06,	0.1126,	210.06,	0.1130,	225.06,	0.1135	A.	341	341		
240.06,	0.1140,	255.06,	0.1146,	270.00,	0.1152	A.	342	342		
285.06,	0.1156,	300.06,	0.1160,	315.06,	0.1163	A.	343	343		
330.06,	0.1166,	345.06,	0.1168,	360.00,	0.1170	A.	344	344		
375.06,	0.1177,	390.06,	0.1174,	405.06,	0.1175	A.	345	345		
420.00,	0.1177,	435.00,	0.1178,	450.00,	0.1178	A.	346	346		
465.12,	0.1179,	540.00,	0.1179			A.	347	347		
555.06,	0.1178,	570.06,	0.1178,	585.06,	0.1178	A.	348	348		
600.00,	0.1177,	END\$				A.	349	349		
108\$ 25C/S, PRIMARY TRIP						A.	350	350		
0.000,	1.0000,	0.200,	1.0495,	0.400,	1.1098	A.	351	351		
0.600,	1.1745,	0.800,	1.2465,	0.810,	0.7618	A.	352	352		
0.855,	0.5953,	0.900,	0.4420,	1.000,	0.3110	A.	353	353		
1.200,	0.1913,	1.400,	0.1478,	1.600,	0.1343	A.	354	354		
2.000,	0.1273,	3.000,	0.1158,	5.000,	0.1023	A.	355	355		
8.000,	.09325,	12.000,	.08325,	END\$		A.	356	356		
208, 0.0, 1.0, 12.0, 1.0, END\$						A.	357	357		
109\$ 25C/S, SECONDARY TRIP						A.	358	358		
0.000,	1.0000,	0.200,	1.0495,	0.400,	1.1098	A.	359	359		
0.600,	1.1745,	0.800,	1.2465,	1.000,	1.3190	A.	360	360		
1.016,	0.9968,	1.1095,	0.9233			A.	361	361		
1.200,	0.8000,	1.400,	0.5018,	1.600,	0.3020	A.	362	362		
2.000,	0.2050,	3.000,	0.1805,	5.000,	0.1545	A.	363	363		
8.000,	0.1325,	12.000,	0.1155,	END\$		A.	364	364		
209, 0.0, 1.0, 12.0, 1.0, END\$						A.	365	365		
110\$ TOTAL LOSS ELECTRIC POWER, - POWER						A.	366	366		
0.0,	1.0000,	1.0,	0.1400,	2.0,	0.1220	A.	367	367		
3.0,	0.1120,	4.0,	0.1050,	5.0,	0.0989	A.	368	368		
10.1,	0.0800,	20.1,	0.0645,	30.0,	0.0566	A.	369	369		
50.1,	0.0474,	100.0,	0.0374,	150.1,	0.0333	A.	370	370		
200.1,	0.0310,	END\$				A.	371	371		
210\$ TOTAL LOSS ELECTRIC POWER, - FLOW						A.	372	372		
0.000,	1.0000,	2.000,	0.7452,	4.093,	0.5813	A.	373	373		
6.045,	0.4782,	8.026,	0.3999,	10.119,	0.3370	A.	374	374		
14.081,	0.2550,	20.152,	0.1809,	24.014,	0.1507	A.	375	375		
30.143,	0.1166,	40.144,	.08051,	50.158,	.05880	A.	376	376		
60.061,	.04532,	70.185,	.03564,	80.128,	.02961	A.	377	377		
90.075,	.02743,	100.021,	.02741,	150.134,	.02966	A.	378	378		
200.056,	.02919,	END\$				A.	379	379		
112\$ PRIMARY PUMP SEIZURE, - POWER						A.	380	380		
0.000,	1.0000,	1.000,	0.2758,	2.000,	0.1216	A.	381	381		
3.001,	0.1114					A.	382	382		
4.001,	0.1042,	5.014,	.09829,	6.041,	.09328	A.	383	383		
7.061,	.08910,	8.025,	.08571,	9.025,	.08267	A.	384	384		

TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL.	1	2	3	4	5	6	7	8	EDIT NO.	CARD NO.
	10.058,	.07995,	15.095,	.07049,	20.027,	.06452			A.	385
	25.134,	.05998,	29.978,	.05659,	35.006,	.05370			A.	386
	40.001,	.05129,	50.123,	.04738,	100.073,	.03742			A.	387
	150.077,	.03328,	200.128,	.03095,	END\$				A.	388
	212\$ PRIMARY PUMP SEIZURE, - FLOW								A.	389
	0.000,	1.0000,	1.000,	0.8244,	2.000,	0.7490			A.	390
	3.001,	0.6510							A.	391
	4.001,	0.5766,	5.014,	0.5153,	6.041,	0.4647			A.	392
	7.061,	0.4230,	8.025,	0.3897,	9.025,	0.3600			A.	393
	10.058,	0.3336,	15.095,	0.2437,	20.027,	0.1911			A.	394
	25.134,	0.1551,	29.978,	0.1307,	35.006,	0.1118			A.	395
	40.001,	.09723,	50.123,	.08238,	100.073,	.08203			A.	396
	150.077,	.08312,	200.128,	.08410,	END\$				A.	397
									A.	398

END

RFL TO 116231B REQUESTED. OLD FL=200000B ,NEW FL=200000B.

ARRAY DIRECTORY

INPUT NO.	STARTING LOCATION	INTERNAL NO.	NO. OF ELEMENTS
2	1	1	13
8	14	2	6
40	20	3	151
41	171	4	151
42	322	5	151
43	473	6	151
44	624	7	151
45	775	8	151
51	926	9	151
52	1077	10	151
53	1228	11	151
54	1379	12	151
55	1530	13	151
61	1681	14	151
62	1832	15	151
63	1983	16	151
64	2134	17	151
65	2285	18	151
66	2436	19	151
71	2587	20	151
72	2738	21	151
73	2889	22	151
74	3040	23	151
75	3191	24	151
101	3342	25	27
201	3369	26	51
102	3420	27	27
202	3447	28	53
103	3500	29	27
203	3527	30	5
104	3532	31	35
204	3567	32	5
105	3572	33	195



## TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

205	3767
106	3944
206	3971
107	3976
207	4011
111	4016
211	4223
108	4410
208	4445
109	4450
209	4483
110	4488
210	4515
112	4554
212	4597

## CARBIDE FUELS PROGRAM

34	177
35	27
36	5
37	35
38	5
39	207
40	187
41	35
42	5
43	33
44	5
45	27
46	39
47	43
48	43

RFL TO 127130B REQUESTED. OLD FL=200000B .NEW FL=200000B.

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO.

CARD NO.

RFL TO 116376B REQUESTED. OLD FL=200000B ,NEW FL=200000B.

	BCD 3EXECUTION	A.	399	399
C	DERIVED QUANTITIES	A.	400	400
	R21 = R99+R11+R11*R14/4.0	A.	401	401
	R22 = R99*(R13+R13-R12+R12)*R14/4.00	A.	402	402
	R27 = R3+R5+R21*(1.0-R1)	A.	403	403
	R28 = R4+R6+R22	A.	404	404
	STFSQS(R27,5,C11)	A.	405	405
	STFSQS(R28/2.,5,C31)	A.	406	406
	STFSQS(R28/2.,5,C41)	A.	407	407
C	AXIAL CONDUCTION	A.	408	408
	R24 = R2+R99*(R13+R13-R12+R12)/8.0/R14	A.	409	409
	STFSQS(R24,4,G131)	A.	410	410
	STFSQS(R24,4,G141)	A.	411	411
F	DO 100 ITEST=2,2	A.	412	412
	MTEST = 0	A.	413	413
	ITERMX = 500	A.	414	414
	JTEST = 0	A.	415	415
	TIMEN = 0.0	A.	416	416
	TIMEND = 0.0	A.	417	417
	TIMED = 0.0	A.	418	418
	STDSTL	A.	419	419
	PRINTA(K97,R21,12,21)	A.	420	420
	WRITE(6,210) ITEST	A.	421	421
	TIMEND = A2+ITEST	A.	422	422
	TSTEPI = .05	A.	423	423
	TSTEPO = 0.2	A.	424	424
	ITERMX = 100	A.	425	425
	JTEST = 1	A.	426	426
	NTEST = 0	A.	427	427
	FWDBCK	A.	428	428
	WRITE(6,200) NTEST	A.	429	429
	WRITE(6,201) (A40+I,A41+I,A42+I,A43+I,A44+I,A45+I,A71+I,A72+I,	A.	430	430
	* A73+I,A74+I,A75+I,A51+I,A52+I,A53+I,A54+I,A55+I,A61+I,A62+I,	A.	431	431
	* A63+I,A64+I,A65+I,A66+I,I=1,NTEST)	A.	432	432
	PUNCH 200, NTEST	A.	433	433
	PUNCH 201, (A40+I,A41+I,A42+I,A43+I,A44+I,A45+I,A71+I,A72+I,	A.	434	434
	* A73+I,A74+I,A75+I,A51+I,A52+I,A53+I,A54+I,A55+I,A61+I,A62+I,	A.	435	435
	* A63+I,A64+I,A65+I,A66+I,I=1,NTEST)	A.	436	436
F 200	FORMAT(I4)	A.	437	437
F 201	FORMAT(X,*TIME = *.F6.2,* (S)+/5X,*POWER (W) *,1X,5F8.1	A.	438	438
F	* /5X,*T FUEL CL (C)*.1X,5F8.1/5X,*T CLAD (C) *,1X,5F8.1	A.	439	439
F	* /5X,*T COOLANT (C)*.1X,5F8.1/5X,*T OUTLET (C) *,1X,5F8.1	A.	440	440
F 210	FORMAT(/X,20H+*****/	A.	441	441
F	*X,*TRANSIENT CASE *.I2/	A.	442	442
F	*X,20H+*****)	A.	443	443
100	CONTINUE	A.	444	444
	END	A.	445	445

## TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

## CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8 EDIT NO.

CARD NO.

	BCD 3VARIABLES1	A.	446	446
	IF(TIMEN.LE.5.0) GO TO 10	A.	447	447
	TSTEPI = 0.05	A.	448	448
	TSTEPO = 0.5	A.	449	449
	IF(TIMEN.LE.10.0) GO TO 10	A.	450	450
	TSTEPO = 1.0	A.	451	451
	IF(TIMEN.LE.50.0) GO TO 10	A.	452	452
	TSTEPI = 0.5	A.	453	453
	TSTEPO = 10.0	A.	454	454
10	CONTINUE	A.	455	455
C	CONDUCTION\IN\FUEL	A.	456	456
	TFUEL1 = R94*T11 + (1.0-R94)*T21	A.	457	457
	TFUEL2 = R94*T12 + (1.0-R94)*T22	A.	458	458
	TFUEL3 = R94*T13 + (1.0-R94)*T23	A.	459	459
	TFUEL4 = R94*T14 + (1.0-R94)*T24	A.	460	460
	TFUEL5 = R94*T15 + (1.0-R94)*T25	A.	461	461
	CFUEL(TFUEL1,R14,R11,R1,R60,R61,R62,R63,R64,GFUEL1,G111)	A.	462	462
	CFUEL(TFUEL2,R14,R11,R1,R60,R61,R62,R63,R64,GFUEL2,G112)	A.	463	463
	CFUEL(TFUEL3,R14,R11,R1,R60,R61,R62,R63,R64,GFUEL3,G113)	A.	464	464
	CFUEL(TFUEL4,R14,R1,R1,R60,R61,R62,R63,R64,GFUEL4,G113)	A.	465	465
	CFUEL(TFUEL5,R14,R11,R1,R60,R61,R62,R63,R64,GFUEL5,G114)	A.	466	466
	DF = 1.0	A.	467	467
	IF(JTEST.EQ.0) DF = R95	A.	468	468
	G11 = DF*GFUEL1 + (1.0-DF)*G11	A.	469	469
	G12 = DF*GFUEL2 + (1.0-DF)*G12	A.	470	470
	G13 = DF*GFUEL3 + (1.0-DF)*G13	A.	471	471
	G14 = DF*GFUEL4 + (1.0-DF)*G14	A.	472	472
	G15 = DF*GFUEL5 + (1.0-DF)*G15	A.	473	473
C	RADIAL CONDUCTION IN CLAD	A.	474	474
	CLAD = 2.0*R99*R14/ALOG(R13/R12)	A.	475	475
	G31 = (R68 + R69*(T31+T41)/2.)*CLAD	A.	476	476
	G32 = (R68 + R69*(T32+T42)/2.)*CLAD	A.	477	477
	G33 = (R68 + R69*(T33+T43)/2.)*CLAD	A.	478	478
	G34 = (R68 + R69*(T34+T44)/2.)*CLAD	A.	479	479
	G35 = (R68 + R69*(T35+T45)/2.)*CLAD	A.	480	480
C	FUEL/CLAD GAP CONDUCTANCE	A.	481	481
	IF(JTEST.GT.0) GO TO 30	A.	482	482
	R111 = T11	A.	483	483
	R112 = T12	A.	484	484
	R113 = T13	A.	485	485
	R114 = T14	A.	486	486
	R115 = T15	A.	487	487
	R121 = T21	A.	488	488
	R122 = T22	A.	489	489
	R123 = T23	A.	490	490
	R124 = T24	A.	491	491
	R125 = T25	A.	492	492
	R131 = T31	A.	493	493
	R132 = T32	A.	494	494
	R133 = T33	A.	495	495
	R134 = T34	A.	496	496
	R135 = T35	A.	497	497

INPUT CARD COL.	= 12345678	1	2345678	2	2345678	3	2345678	4	2345678	5	2345678	6	2345678	7	2345678	8	EDIT NO.	CARD NO.	
			R141 = T41														A.	498	498
			R142 = T42														A.	499	499
			R143 = T43														A.	500	500
			R144 = T44														A.	501	501
			R145 = T45														A.	502	502
			R155 = T55														A.	503	503
	30		HGAP(1,T11,T21,R111,R121,T31,T41,R131,R141,T55,R155,R11,R13,														A.	504	504
		*	R12,RB,R1,R41,R71)														A.	505	505
			HGAP(1,T12,T22,R112,R122,T32,T42,R132,R142,T55,R155,R11,R13,														A.	506	506
		*	R12,RB,R1,R42,R72)														A.	507	507
			HGAP(1,T13,T23,R113,R123,T33,T43,R133,R143,T55,R155,R11,R13,														A.	508	508
		*	R12,RB,R1,R43,R73)														A.	509	509
			HGAP(1,T14,T24,R114,R124,T34,T44,R134,R144,T55,R155,R11,R13,														A.	510	510
		*	R12,RB,R1,R44,R74)														A.	511	511
			HGAP(1,T15,T25,R115,R125,T35,T45,R135,R145,T55,R155,R11,R13,														A.	512	512
		*	R12,RB,R1,R45,R75)														A.	513	513
			R29 = R99*R11*R14														A.	514	514
			G21 = (1.0-R90)*G21 + R90*R29*R71														A.	515	515
			G22 = (1.0-R90)*G22 + R90*R29*R72														A.	516	516
			G23 = (1.0-R90)*G23 + R90*R29*R73														A.	517	517
			G24 = (1.0-R90)*G24 + R90*R29*R74														A.	518	518
			G25 = (1.0-R90)*G25 + R90*R29*R75														A.	519	519
	C		COOLANT FLOWRATE AND POWER														A.	520	520
			GO TO (11,12,13,14,15,16,17,18,19,20,21,22),ITEST														A.	521	521
	11		D1D1WM(TIMEN,A101,R15,R30)														A.	522	522
			D1D1WM(TIMEN,A201,R9,R31)														A.	523	523
			GO TO 99														A.	524	524
	12		D1D1WM(TIMEN,A102,R15,R30)														A.	525	525
			D1D1WM(TIMEN,A202,R9,R31)														A.	526	526
			GO TO 99														A.	527	527
	13		D1D1WM(TIMEN,A103,R15,R30)														A.	528	528
			D1D1WM(TIMEN,A203,R9,R31)														A.	529	529
			GO TO 99														A.	530	530
	14		D1D1WM(TIMEN,A104,R15,R30)														A.	531	531
			D1D1WM(TIMEN,A204,R9,R31)														A.	532	532
			GO TO 99														A.	533	533
	15		D1D1WM(TIMEN,A105,R15,R30)														A.	534	534
			D1D1WM(TIMEN,A205,R9,R31)														A.	535	535
			R30 = 0.93*R30														A.	536	536
			R31 = 0.93*R31														A.	537	537
			GO TO 99														A.	538	538
	16		D1D1WM(TIMEN,A106,R15,R30)														A.	539	539
			D1D1WM(TIMEN,A206,R9,R31)														A.	540	540
			GO TO 99														A.	541	541
	17		D1D1WM(TIMEN,A107,R15,R30)														A.	542	542
			D1D1WM(TIMEN,A207,R9,R31)														A.	543	543
			GO TO 99														A.	544	544
	18		D1D1WM(TIMEN,A108,R15,R30)														A.	545	545
			D1D1WM(TIMEN,A208,R9,R31)														A.	546	546
			GO TO 99														A.	547	547
	19		D1D1WM(TIMEN,A109,R15,R30)														A.	548	548
			D1D1WM(TIMEN,A209,R9,R31)														A.	549	549

## TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

## CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD COL. = 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678.8 EDIT NO.

CARD NO.

	GO TO 99	A.	550	550
20	D1D1WM(TIMEN,A110,R15,R30)	A.	551	551
	D1D1WM(TIMEN,A210,R9,R31)	A.	552	552
	GO TO 99	A.	553	553
21	D1D1WM(TIMEN,A111,R15,R30)	A.	554	554
	D1D1WM(TIMEN,A211,R9,R31)	A.	555	555
	R30 = 0.93*R30	A.	556	556
	R31 = 0.93*R31	A.	557	557
	GO TO 99	A.	558	558
22	D1D1WM(TIMEN,A112,R15,R30)	A.	559	559
	D1D1WM(TIMEN,A212,R9,R31)	A.	560	560
	GO TO 99	A.	561	561
99	CONTINUE	A.	562	562
	R32 = R31*R16	A.	563	563
C	HEAT GENERATION IN PIN	A.	564	564
	Q11 = R7*R30*R21*R51	A.	565	565
	Q12 = R7*R30*R21*R52	A.	566	566
	Q13 = R7*R30*R21*R53	A.	567	567
	Q14 = R7*R30*R21*R54	A.	568	568
	Q15 = R7*R30*R21*R55	A.	569	569
	A8+1 = Q11	A.	570	570
	A8+2 = Q12	A.	571	571
	A8+3 = Q13	A.	572	572
	A8+4 = Q14	A.	573	573
	A8+5 = Q15	A.	574	574
C	FLUID NETWORK	A.	575	575
	FILM(T50,T51,R32,R17,R18,R13,R14,G41,G61,G51)	A.	576	576
	FILM(T51,T52,R32,R17,R18,R13,R14,G42,G62,G52)	A.	577	577
	FILM(T52,T53,R32,R17,R18,R13,R14,G43,G63,G53)	A.	578	578
	FILM(T53,T54,R32,R17,R18,R13,R14,G44,G64,G54)	A.	579	579
	FILM(T54,T55,R32,R17,R18,R13,R14,G45,G65,G55)	A.	580	580
	END	A.	581	581

INPUT CARD COL.	= 12345678	1	2345678	2	2345678	3	2345678	4	2345678	5	2345678	6	2345678	7	2345678	8	EDIT NO.	CARD NO.	
																	A.	582	582
																	A.	583	583
																	A.	584	584
																	A.	585	585
																	A.	586	586
																	A.	587	587
																	A.	588	588
																	A.	589	589
																	A.	590	590
																	A.	591	591
																	A.	592	592
																	A.	593	593
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																	A.	612	612
																	A.	613	613
																	A.	614	614
																	A.	615	615
																	A.	616	616
																	A.	617	617
																	A.	618	618
																	A.	619	619
																	A.	620	620
																	A.	621	621
																	A.	622	622

## TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

## CARBIDE FUELS PROGRAM

LANL MAY 198

INPUT CARD CDL.	= 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8	EDIT NO.	CARD NO.
	BCD 3SUBROUTINES	A.	623
F	SUBROUTINE HGAP(IFLAG,TFI,TFD,TFSSI,TFSSO,TCI,TCQ,TCSSI,TCSSO,	A.	624
F	\$ TFLO,TFLOSS,DFUEL,DCLADO,DCLADI,PNOT,PORE,PINTF,GGAP)	A.	625
C		A.	626
C	THIS SUBROUTINE COMPUTES THE GAP CONDUCTANCE BETWEEN CLADDING AND	A.	627
C	FUEL FOR BOTH SPHERE-PAC AND PELLET FUEL PINS.	A.	628
C	AN "ADAPTED" ROSS/STOUTE METHOD IS USED	A.	629
C	WHERE THE DEPENDENCE OF CONDUCTANCE ON INTERFACE PRESSURE IS A	A.	630
C	CURVE FIT BASED ON LIFE-4C COMPUTATIONS.	A.	631
C		A.	632
C	CONSTANTS FOR D9 CLADDING	A.	633
C		A.	634
F	DATA AEC,BEC/2.01E+05,-79.29/	A.	635
F	DATA AGC,BGC/7.86E+04,-38.41/	A.	636
F	DATA AAC,BAC/15.E-06,4.E-09/	A.	637
C		A.	638
C	CONSTANTS FOR CARBIDE FUEL	A.	639
C		A.	640
F	DATA AAF,BAF/9.,1E-06,1.4E-09/	A.	641
C		A.	642
C	DATA FOR PLENUM IC'S	A.	643
C		A.	644
F	DATA PPNOT,TPNOT/330000.,871./	A.	645
C		A.	646
C	CALCULATE ELASTIC CONSTANTS	A.	647
C		A.	648
F	TCLAD=0.5*(TCI+TCQ)	A.	649
F	ECLAD=(AEC+BEC*TCLAD)*1.E+6	A.	650
F	GCLAD=(AGC+BGC*TCLAD)*1.E+6	A.	651
F	ALPHC=AAC+BAC*TCLAD	A.	652
F	GNUC=(ECLAD/(2.*GCLAD))-1.	A.	653
C		A.	654
F	TFUEL=0.5*(TFI+TFD)	A.	655
F	EFUEL=2.02*(1.-1.54*PORE)*(1.-0.92E-04*(TFUEL-25.))*1.E+11	A.	656
F	GFUEL=0.78*(1.-1.54*PORE)*(1.-0.92E-04*(TFUEL-25.))/(1.-0.44E-04*	A.	657
F	(TFUEL-25.))*1.E+11	A.	658
F	\$ ALPHF=AAF+BAF*TFUEL	A.	659
F	GNUF=(EFUEL/(2.*GFUEL))-1.	A.	660
C		A.	661
C	CALCULATE CHANGE IN PLENUM PRESSURE	A.	662
C		A.	663
F	DELPP=PPNOT*((TFLO+273.)/TPNOT-1.0)	A.	664
C		A.	665
C	CALCULATE INTERFERANCE BASED ON NEW TEMPERATURES	A.	666
C	AND PLENUM PRESSURES	A.	667
C		A.	668
F	TICLAD=0.5*(TCSSI+TCSSO)	A.	669
F	TIFUEL=0.5*(TFSSI+TFSSO)	A.	670
F	RCI=(DCLADI/2.)/100.	A.	671
F	RCD=(DCLADO/2.)/100.	A.	672
F	THIKCL=RCD-RCI	A.	673
F	RF=(DFUEL/2.)/100.	A.	674

INPUT CARD COL.	= 12345678 1 2345678 2 2345678 3 2345678 4 2345678 5 2345678 6 2345678 7 2345678 8	EDIT NO.	CARD NO.
F	DELTA=((1.+GNUF)*ALPHF*RF*(TFUEL-TIFUEL)-ALPHC*RCI*(TCLAD-TICLAD)	A.	675
F	%) -DELPP*(RCI*RCI/(2.*ECLAD*THIKCL))	A.	676
C		A.	677
C	CALCULATE INTERFACE PRESSURE BASED ON NEW DELTA	A.	678
C		A.	679
F	DENOM=RCI*(1./EFUEL+(RCD*RCD+RCI*RCI)/(ECLAD*(RCD*RCD-RCI*RCI))	A.	680
F	\$ -GNUF/EFUEL+GNUF/ECLAD)	A.	681
F	PINTF=DELTA/DENOM+PNOT	A.	682
C		A.	683
C	CALCULATE GAP CONDUCTANCE	A.	684
C		A.	685
F	IF(PINTF.LE.O.O) PINTF=O.O	A.	686
F	GGAP=1.005+5.218*((PINTF+1.E-06)/18.63)**1.15	A.	687
C		A.	688
F	RETURN	A.	689
F	END	A.	690
F	SUBROUTINE FILM(TF1,TF2,MCP,DCHAN,ACHAN,DCLAD,LEN,G1,G2,G3)	A.	691
F	REAL MCP,LEN	A.	692
F	PI = 3.14159	A.	693
F	ASURF = PI*DCLAD*LEN	A.	694
F	TFM = (TF1+TF2)/2.O	A.	695
F	COND = 0.9291 + TFM*(1.1723E-7+TFM-5.8064E-4)	A.	696
F	PECLET = MCP/2.*DCHAN/ACHAN/COND	A.	697
F	HTC = COND/DCHAN*(0.012+PECLET**.86+4.386)	A.	698
F	UA = HTC*ASURF	A.	699
F	IF(UA/MCP.GT.1.OE3) GD TO 10	A.	700
F	EFF = 1.O-EXP(-UA/MCP)	A.	701
F	G1 = EFF*MCP	A.	702
F	G2 = G1	A.	703
F	G3 = (1.O-EFF)*MCP	A.	704
F	RETURN	A.	705
F	10 G1 = MCP	A.	706
F	G2 = G1	A.	707
F	G3 = 1.OE-3	A.	708
F	RETURN	A.	709
F	END	A.	710
F	SUBROUTINE CFUEL(TFUEL,LEN,DFUEL,PDRO,CO,C1,C2,C3,C4,	A.	711
F	* CONDR,CONDA)	A.	712
C	THIS SUBROUTINE CALCULATES FUEL CONDUCTANCE, AXIAL AND RADIAL,	A.	713
C	AS A FUNCTION OF TEMPERATURE UNITS ARE W/CM-K FOR T IN KELVINS	A.	714
F	REAL LEN,KFUEL	A.	715
F	TFUEL = TFUEL + 273.15	A.	716
F	KFUEL = (CO + TFUEL*(C1 + TFUEL*(C2 + TFUEL*(C3 + TFUEL*C4))))	A.	717
F	IF(KFUEL.GT.O.15) KFUEL = O.15	A.	718
F	CONDR = 4.O*KFUEL*3.14159*LEN	A.	719
F	CONDA = KFUEL*3.14159*DFUEL*DFUEL/4.O/LEN	A.	720
F	RETURN	A.	721
F	END	A.	722
F	BCD 3END OF DATA	A.	723

RFL TO 127050B REQUESTED. OLD FL=200000B ,NEW FL=200000B.



TRANSPA: TRANSIENT SINGLE PIN ANALYSIS

CARBIDE FUELS PRDGRAM

THE INITIAL DATA VALUES AND FORTRAN PROGRAMS ( AFTER EDITS) HAVE BEEN  
WRITTEN ON THE PROBLEM RESTART FILE -RSTRTO- AS PROBLEM-STOOOO-.

CORE REQUIREMENTS FOR MITAS COMPILATION PHASE		
NODE BLOCK.....	0116127	OCTAL
CONDUCTOR BLOCK.....	0116147	OCTAL
CONSTANTS BLOCK.....	0116374	OCTAL
ARRAY BLOCK.....	0127130	OCTAL
VARIABLE CAP. AND/OR COND. ARRAY CONSTRUCTION.....	0000000	OCTAL
OPERATIONS BLOCK INTERPRETATION.....	0116376	OCTAL
FILE SETUP.....	0127050	OCTAL
PARAMETRIC AND/OR RESTART.....	0000000	OCTAL
 MINIMUM CORE REQUIRED FOR MITAS COMPILER.....	 0127200	 OCTAL

CP TIME CONSUMED IN MITAS COMPILATION PHASE		
EDIT PROCESSING.....		.567 SEC
NODE BLOCK.....	.229	
CONDUCTOR BLOCK.....	.306	
CONSTANT BLOCK.....	.322	
ARRAY BLOCK.....	1.170	
VARIABLE CAP. AND/OR CONO.....	.052	
TOTAL TIME FOR DATA BLOCKS.....		2.115 SEC
TOTAL TIME FOR OPERATIONS BLOCK.....		1.681 SEC
TOTAL TIME FOR FILE SETUP.....		0.000 SEC
TOTAL TIME FOR PARAMETRIC AND/OR RESTART.....		0.000 SEC
		-----
TOTAL CP TIME FOR MITAS COMPILATION PHASE.....		4.847 SEC

+++++ NORMAL TERMINATION OF MITAS COMPILER +++++